- We have all seen moving fluids exerting forces. The lift force on an aircraft is exerted by the air moving over the wing. A jet of water from a hose exerts a force on whatever it hits.
- In fluid mechanics the analysis of motion is performed in the same way as in solid mechanics - by use of Newton's laws of motion.
- i.e., F = ma which is used in the analysis of solid mechanics to relate applied force to acceleration.
- In fluid mechanics it is not clear what mass of moving fluid we should use so we use a different form of the equation.

$$\sum F = ma = \frac{d(mV)_s}{dt}$$

- Newton's 2nd Law can be written:
- The Rate of change of momentum of a body is equal to the resultant force acting on the body, and takes place in the direction of the force.

$$\sum \mathbf{F} = \frac{d(m\mathbf{V})_x}{dt}$$

 $\sum F$ = Sum of all external forces on a body of fluid or system s

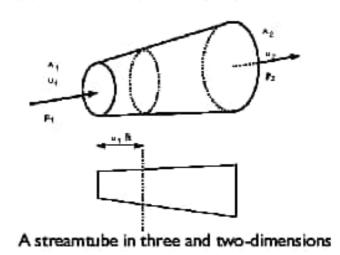
mV = Momentum of fluid body in direction s

 The symbols F and V represent vectors and so the change in momentum must be in the same direction as force.

$$\sum Fdt = d(mV)_s$$

22 It is also termed as impulse momentum principle

- Let's start by assuming that we have steady flow which is non-uniform flowing in a stream tube.
- In time δt a volume of the fluid moves from the inlet a distance u δt, so the volume entering the streamtube in the time δt is



volume entering the stream tube = area x distance = $A_1u_1\delta t$ mass entering stream tube = volume x density = $\rho_1A_1u_1\delta t$ momentum of fluid entering stream tube = mass x velocity= $(\rho_1A_1u_1\delta t)u_1$ momentum of fluid leaving stream tube = $(\rho_2A_2u_2\delta t)u_2$

- Now, according to Newton's 2nd Law the force exerted by the fluid is equal to the rate of change of momentum. So
- Force=rate of change of momentum

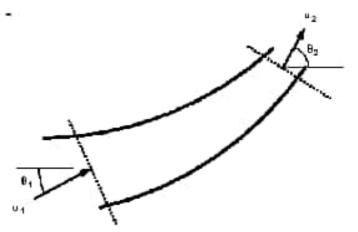
$$\sum \mathbf{F} = \frac{\rho_2 A_2 u_2 \delta t u_2 - \rho_1 A_1 u_1 \delta t u_1}{\delta t} = \frac{\rho_2 A_2 u_2 \delta t u_2}{\delta t} - \frac{\rho_1 A_1 u_1 \delta t u_1}{\delta t}$$
$$\sum \mathbf{F} = \frac{\rho_2 (A_2 u_2 \delta t) u_2}{\delta t} - \frac{\rho_1 (A_1 u_1 \delta t) u_1}{\delta t} = \rho_2 (Q_2) u_2 - \rho_1 (Q_1) u_1$$

We know from continuity of incompressible flow, ρ=ρ₁= ρ₂ & Q=Q₁=Q₂

$$F = \rho Q[u_2 - u_1] = m[u_2 - u_1]$$

This analysis assumed that the inlet and outlet velocities were in the same direction - i.e. a one dimensional system. What happens when this is not the case?

- Consider the two dimensional system in the figure below:
- At the inlet the velocity vector, ul, makes an angle, θl, with the x-axis, while at the outlet u2 make an angle θ 2.
- In this case we consider the forces by resolving in the directions of the co-ordinate axes.



Two dimensional flow in a streamtube

The force in the x-direction

 F_* = Rate of change of momentum in x - direction

= Rate of change of mass × change in velocity in x-direction

$$= \dot{m}(u_1 \cos \theta_1 - u_1 \cos \theta_1)$$

$$= \dot{m}(u_{1_{\bullet}} - u_{1_{\bullet}})$$

$$= \rho Q(u_2 \cos \theta_2 - u_1 \cos \theta_1)$$

$$= \rho Q(u_2, -u_1)$$

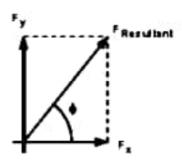
The force in the y-direction

$$F_{y} = in(u_{2} \sin \theta_{2} - u_{1} \sin \theta_{1})$$

$$= in(u_{2,y} - u_{1,y})$$

$$= \rho Q(u_{2} \sin \theta_{2} - u_{1} \sin \theta_{1})$$

$$= \rho Q(u_{2,y} - u_{1,y})$$



 The resultant force can be determined by combining Fx and Fy vectorially as

$$F_{\text{resultant}} = \sqrt{F_s^2 + F_s^2}$$

And the angle at which Facts is given by

$$\phi = \tan^{-1}\left(\frac{F_r}{F_r}\right)$$

- For a three-dimensional (x, y, z) system we then have an extra force to calculate and resolve in the z direction.
- This is considered in exactly the same way.
- In summary we can say: The total force the fluid = rate of change of momentum through the control volume

$$F = \dot{m}(u_{\text{out}} - u_{\text{in}})$$
$$= \rho Q(u_{\text{out}} - u_{\text{in}})$$

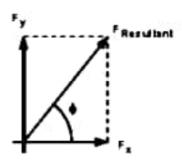
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$$F = \dot{m}(u_{\text{out}} - u_{\text{in}})$$
$$= \rho Q(u_{\text{out}} - u_{\text{in}})$$

- Note that we are working with vectors so F is in the direction of the velocity. This force is made up of three components:
- F_R = Force exerted on the fluid by any solid body touching the control volume
- F_B = Force exerted on the fluid body (e.g. gravity)
- F_p = Force exerted on the fluid by fluid pressure outside the control volume
- So we say that the total force, F_T is given by the sum of these forces:

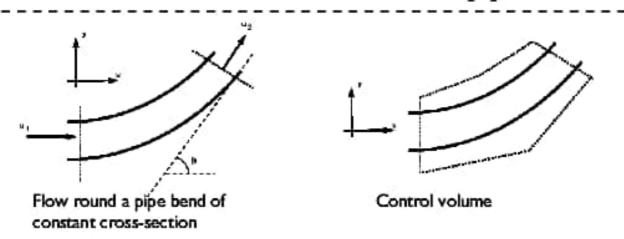
$$F_T = F_R + F_B + F_P$$

 The force exerted by the fluid on the solid body touching the control volume is opposite to F_R. So the reaction force, R, is given by

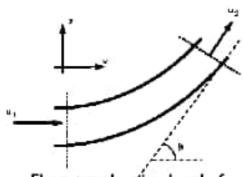
 $R = F_p$

Application of the Momentum Equation

- In common application of the momentum principle, we use it to find forces that flowing fluid exert on structures open to the atmosphere like gate and overflow spillways
- In the following section, we will consider the application of momentum principle for the following cases.
 - I. Force due to the flow of fluid round a pipe bend.
 - 2. Force on a nozzle at the outlet of a pipe.
 - 3. Impact of a jet on a plane surface.
 - 4. Force due to flow round a curved vane.



- Coordinate system: It is convenient to choose the co-ordinate axis so that one is pointing in the direction of the inlet velocity.
- In the above figure the x-axis points in the direction of the inlet velocity.
- Let's compute, total force, pressure force, body force and resultant force



Flow round a pipe bend of constant cross-section



Control volume

I.Total Force:

In x-direction

$$F_{r_*} = \rho Q(u_{r_*} - u_{r_*})$$

$$u_{r_*} = u_{r_*}$$

$$u_{r_*} = u_{r_*} \cos \theta$$

$$F_{r_*} = \rho Q(u_{r_*} \cos \theta - u_{r_*})$$

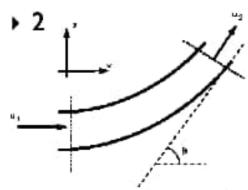
In y-direction

$$F_{T,y} = \rho Q(u_{2,y} - u_{1,y})$$

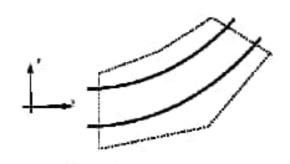
$$u_{1,y} = u_1 \sin \theta = 0$$

$$u_{2,y} = u_2 \sin \theta$$

$$F_{T,y} = \rho Q u_2 \sin \theta$$



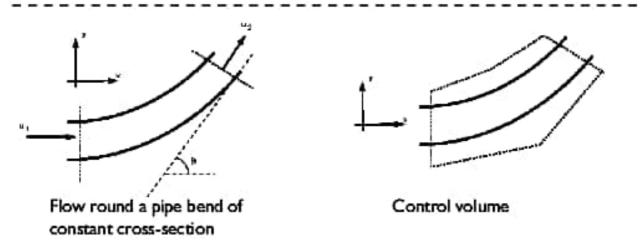
Flow round a pipe bend of constant cross-section



Control volume

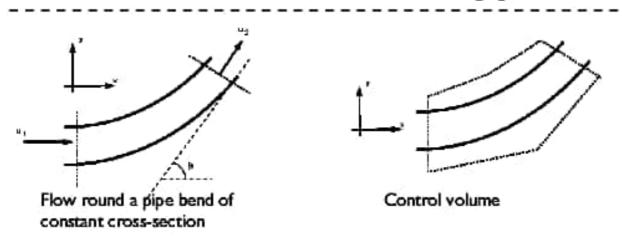
2. Pressure force

$$F_p$$
 = pressure force at 1 - pressure force at 2
 $F_{P_x} = p_1 A_1 \cos 0 - p_2 A_2 \cos \theta = p_1 A_1 - p_2 A_2 \cos \theta$
 $F_{P_y} = p_1 A_1 \sin 0 - p_2 A_2 \sin \theta = -p_2 A_2 \sin \theta$



3. Body force:

There are no body forces in the x or y directions. The only body force is that exerted by gravity (which acts into the paper in this example - a direction we do not need to consider).



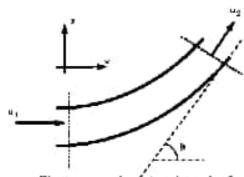
Resultant force

$$F_{T_x} = F_{R_x} + F_{P_x} + F_{R_x}$$

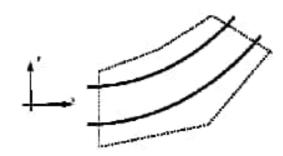
$$F_{T_y} = F_{R_y} + F_{P_y} + F_{R_y}$$

$$F_{R_x} = F_{T_x} - F_{P_x} - 0 = \rho Q(u_2 \cos \theta - u_1) - \rho_1 A_1 + \rho_2 A_2 \cos \theta$$

$$F_{R_y} = F_{T_y} - F_{P_y} - 0 = \rho Q u_2 \sin \theta + \rho_2 A_2 \sin \theta$$



Flow round a pipe bend of constant cross-section

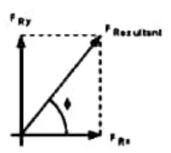


Control volume

Resultant force and direction

$$F_R = \sqrt{F_{R_x}^2 - F_{R_y}^2}$$
 $\phi = \tan^{-1} \left(\frac{F_{R_y}}{F_{R_x}} \right)$

$$\phi = \tan^{-1} \left(\frac{F_{R_s}}{F_{R_s}} \right)$$



 Finally, the force on bent is same magnitude but in opposite direction

$$R = -F_R$$